

PATENT  
Docket 22NORDAM28

In the Specification

Please substitute the following paragraphs for the corresponding paragraphs being amended:

[0047] A corresponding radially inner reverser or blocker door 56 is suitably pivotally joined to the fan nacelle 16 inside the compartment 50 in radial opposition with the gang of louver doors 54 to ~~the~~ close the inlet end of the tunnel along the inner skin 40. In the stowed closed position illustrated in Figure 4, the inner door 56 is folded closed generally parallel with the corresponding gang of outer doors 54, converging slightly to conform with the converging profile or cross section of the nacelle.

[0087] The three blocker ~~door~~ doors 56 in each nacelle half preferably have trapezoidal configurations for circumferentially adjoining each other inside the inner skin 40 when deployed as illustrated in Figure 3 for blocking the fan flow from exiting through the fan nozzle. Instead, the fan flow is diverted through the open louver doors as illustrated in Figure 2 and directed axially forwardly for providing thrust reverse operation in landing of the aircraft.